

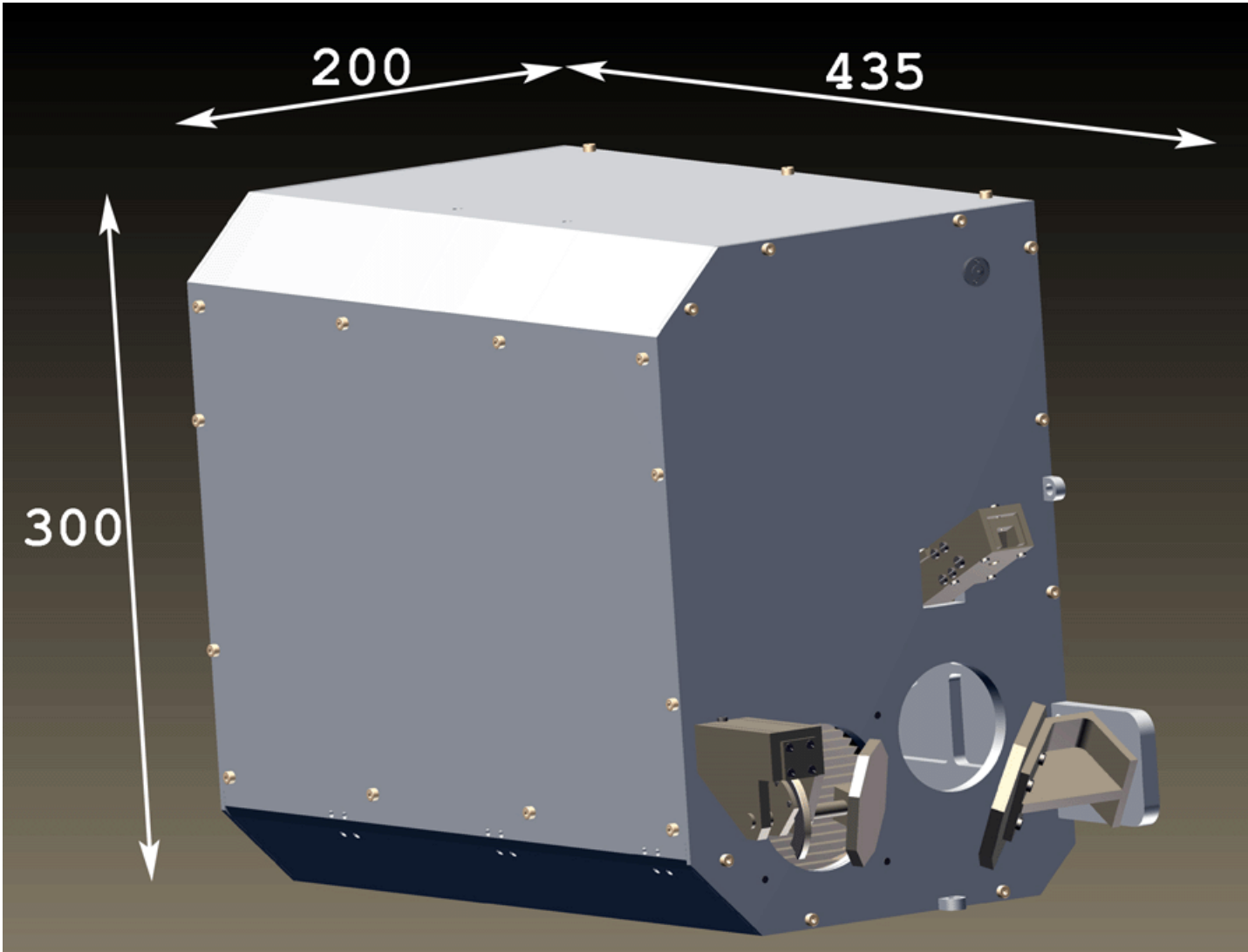
# Fourier-spectrometers in ExoMars missions:

TIRVIM/ACS (2016)  
and  
FAST (2018)

# Fourier-spectrometers in ExoMars missions

MISSION	EXPERIMENT	STATUS
Trace Gases Orbiter, “ExoMars-2016” (2 <sup>h</sup> polar orbit)	<b>TIRVIM/ACS:</b> 0.2cm <sup>-1</sup> ; 2-17μ; 12kg; 2”-aperture; single-direction scanner	Is being funded by Roscosmos
Landing Platform, “ExoMars-2018”	<b>FAST:</b> 0.05cm <sup>-1</sup> ; 2-17μ; 4kg; 1”-aperture; bi-directional scanner	Got the top grade at contest, the funding is pending

## Instrument overview



## Main scientific objectives

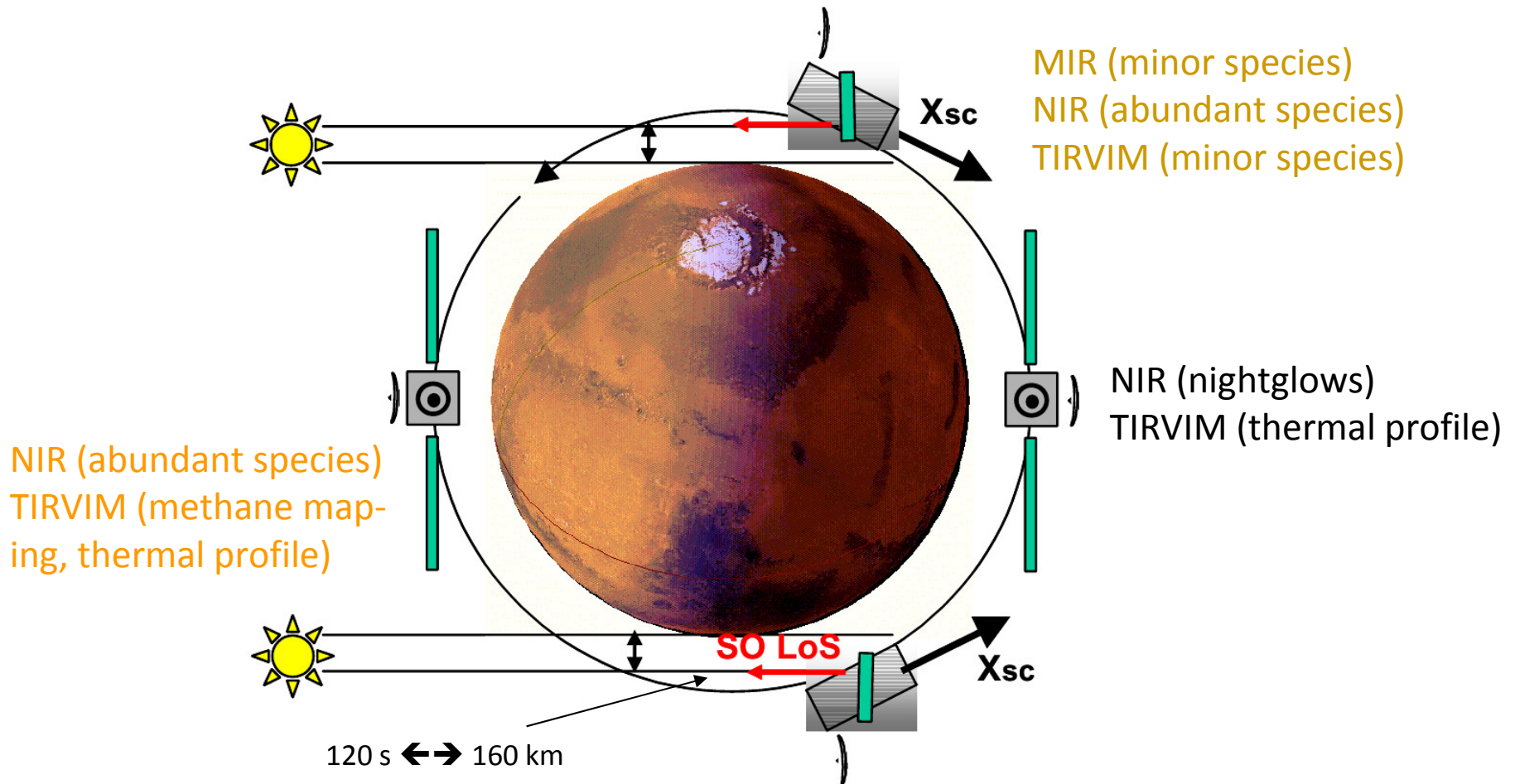
OBJECTIVE	MESUREMENT MODES & PARAMETERS
Methane & other minor atmosphere constituents	Sun occultations, 2-17 $\mu$ m, 0.2 $\text{cm}^{-1}$ 1) PV-MCT detector @65K, 1 IFG: 2sec, S/N $\sim$ 10 <sup>3</sup> or (redundant channel): 2) Pyro-detector @RT, 1 IFG: 30sec, S/N $\sim$ 3x10 <sup>2</sup>
Methane, if any, mapping at day-side	Nadir, 2-4 $\mu$ m, 0.2 $\text{cm}^{-1}$ PV-PbCdSe detector @200K, 1 IFG: 10sec, S/N $\sim$ 3x10 <sup>2</sup>
Vertical thermal profile of the atmosphere, both day-side & night- side	Nadir, CO <sub>2</sub> band at 15 $\mu$ m, 1.6 $\text{cm}^{-1}$ PV-MCT detector @65K, 1 IFG: 4sec, S/N $\sim$ 3x10 <sup>2</sup>

On-board FFT and scissor mode apply to all modes

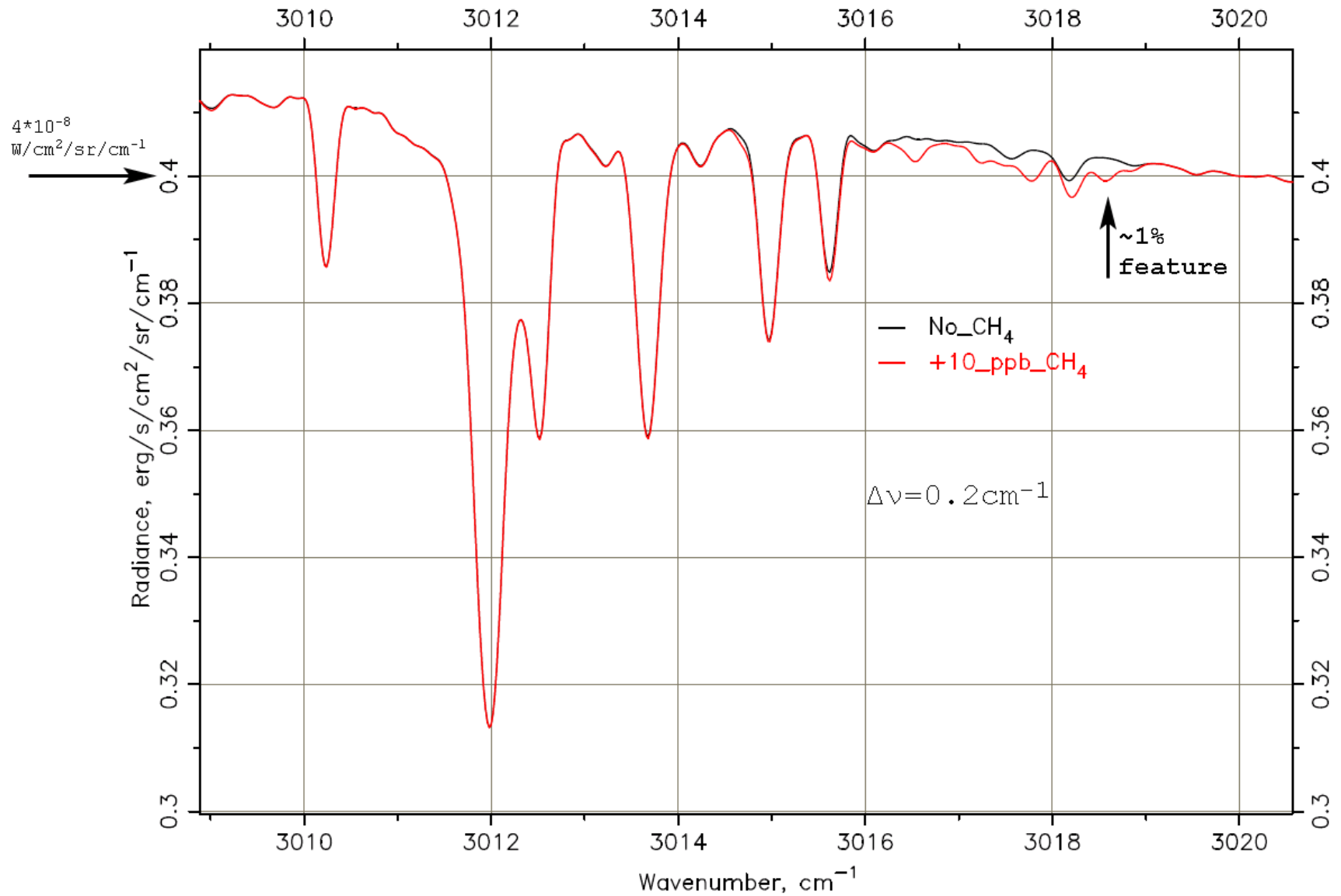
# Experiment Operation Plan

TGO operational orbit  $T \sim 2$  hr orbital period

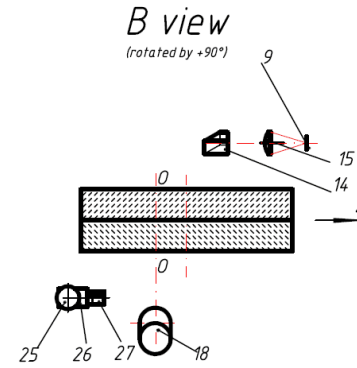
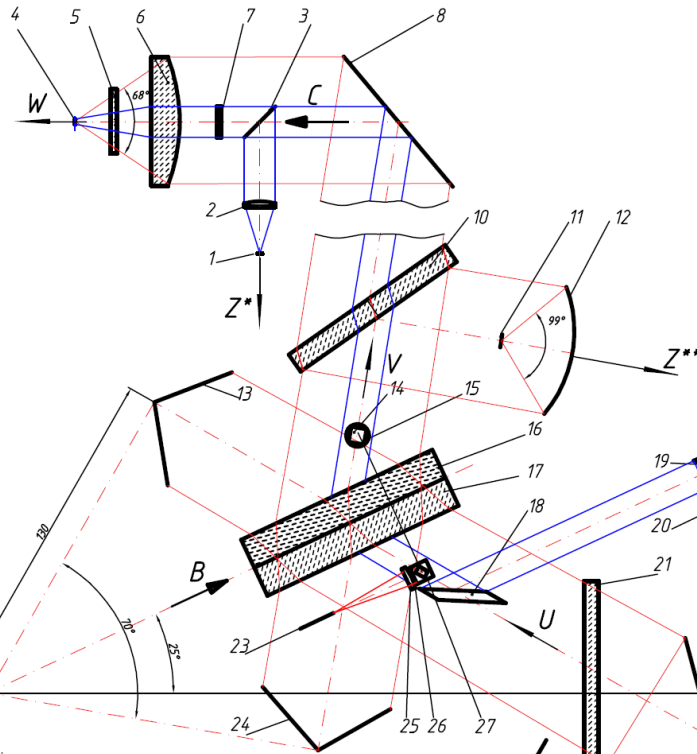
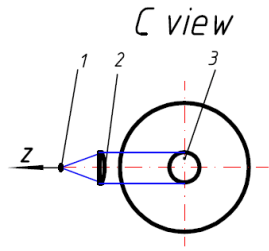
→ 12 orbits per day, 12 sunrises and 12 sunsets – 24 occultation's per day



# CH<sub>4</sub> simulated spectra (nadir)



## Optical scheme



- 1 Pyro-detector
- 2 Pyro-detector' lens (ZnSe)
- 3 Flip mirror (flat)
- 4 PV-MCT detector (@65K)
- 5 ARC-Ge window (2.5mm thick)
- 6 Aspheric ZnSe lens (Ø50)
- 7 Diverging ZnSe lens (Ø12 R=-150)
- 8 Fixed flat mirror (CA: 50x66)
- 9 Reference Channel (RC) detector
- 10 Dichroic plate
- 11 PV-PbCdSe detector (@200K)
- 12 Parabolic mirror (Ø50 R=50)
- 13 Retro-reflector (CA=Ø51)
- 14 RC-detector mirroring prism
- 15 RC-detector lens
- 16 Compensator (ZnSe Ø88x12)
- 17 Beam-splitter (ZnSe Ø88x12)
- 18 Flat mirror(Sun periscope outlet)
- 19 Sun-inlet mirroring prism
- 20 Sun-inlet filter & field stop
- 21 Nadir-inlet window (TBD)
- 22 Fixed flat mirror (CA: 55x78)
- 23 RC-emitter fiber optic
- 24 Retro-reflector (CA=Ø51)
- 25 RC-emitter collimating lens
- 26 RC-emitter mirroring-prism-big
- 27 RC-emitter mirroring-prism-small
- 28 Calibration Black-Body
- 29 Scanning flat mirror (CA: 58x84)

Mirror 29 scans by rotation around Z axis.

00 - optical axis of the Interferometer.

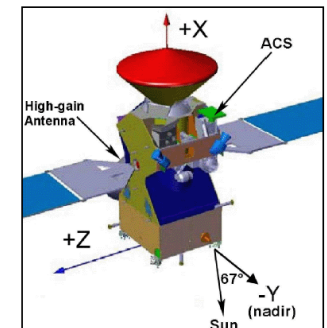
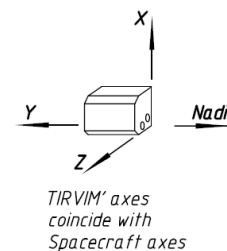
$Y^*$ ,  $Z^*$ ,  $Z^{**}$  &  $Z^{***}$  - Instrument' axes after conventional rotations.

Elements 1, 2 & 3 at the main view are conventionally rotated by -90° around axis W.

Elements 10, 11 & 12 are conventionally rotated by +90° around axis V.

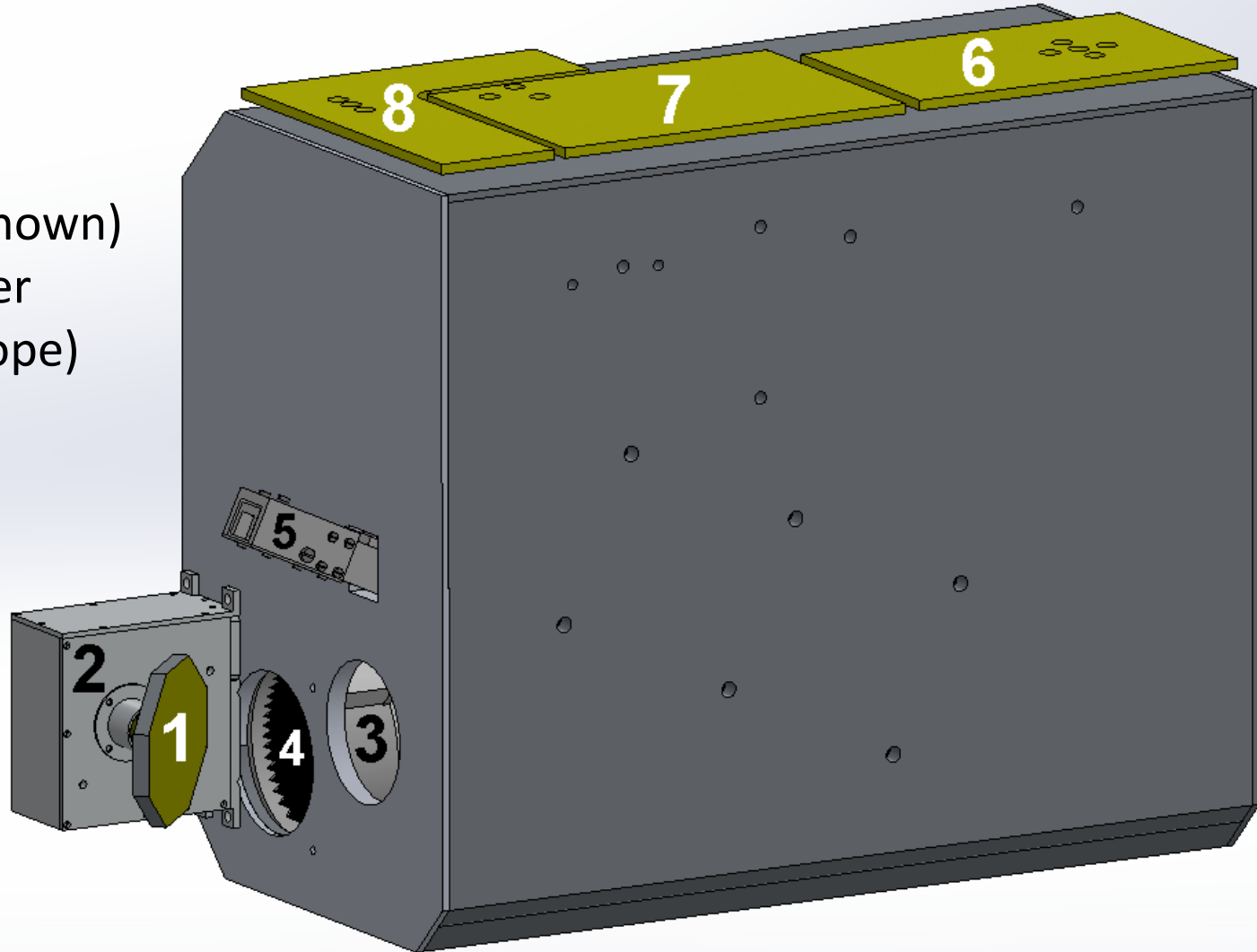
Elements 22, 28 & 29 are conventionally rotated by -90° around axis U, and then elements 28 & 29 are additionally conventionally rotated by -30° around axis  $Z^{***}$ .

Elements 1..8 are conventionally moved by 30mm along the V axis.



## Instrument overview continued

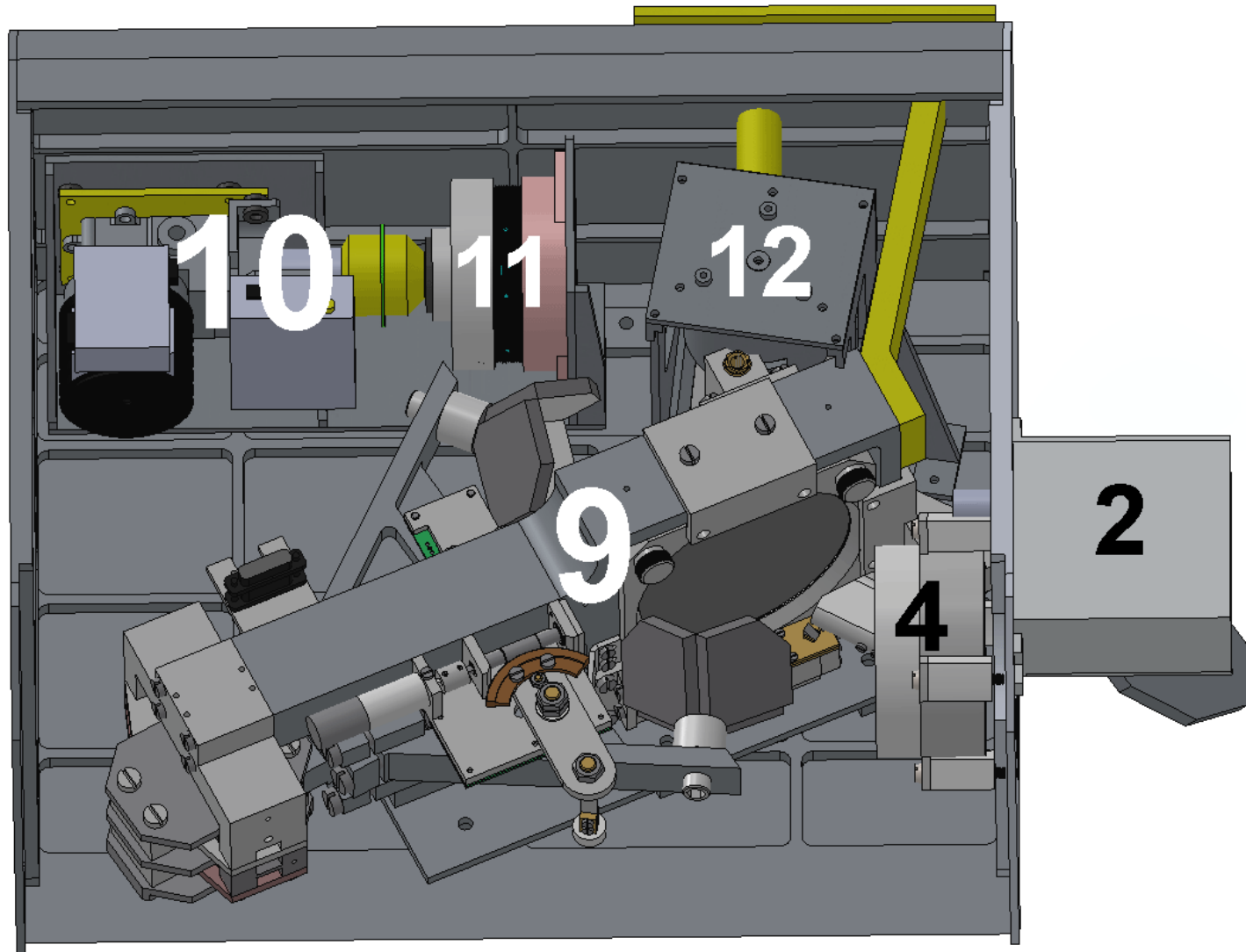
- 1 – Scanning mirror
- 2 – Scanner module
- 3 – Optical inlet  
(fixed mirror not shown)
- 4 – Blackbody emitter
- 5 – Sun inlet (periscope)
- 6 – Stirling radiator
- 7 – PbCdSe radiator
- 8 – Interferometer radiator



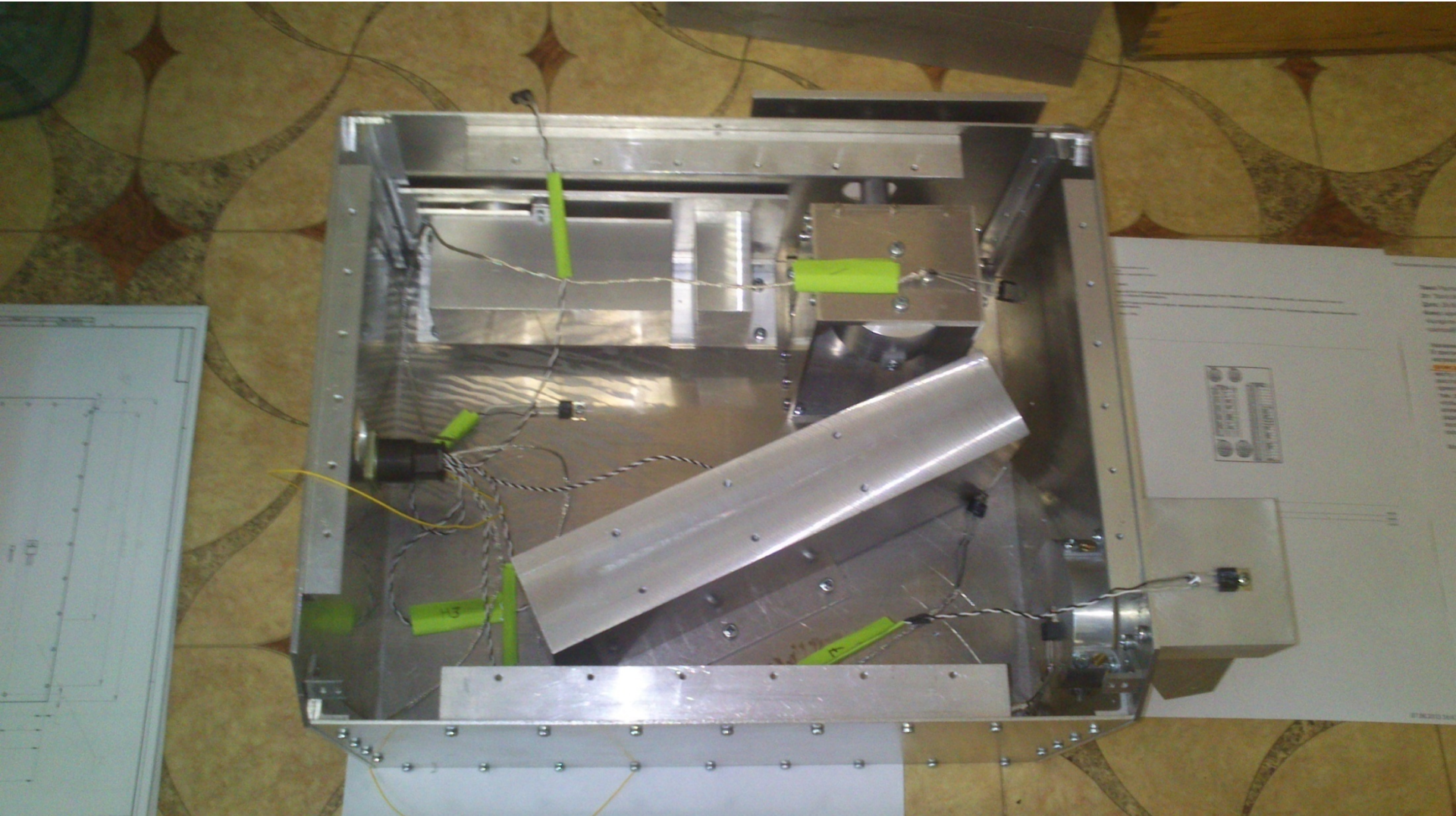


## Instrument overview continued

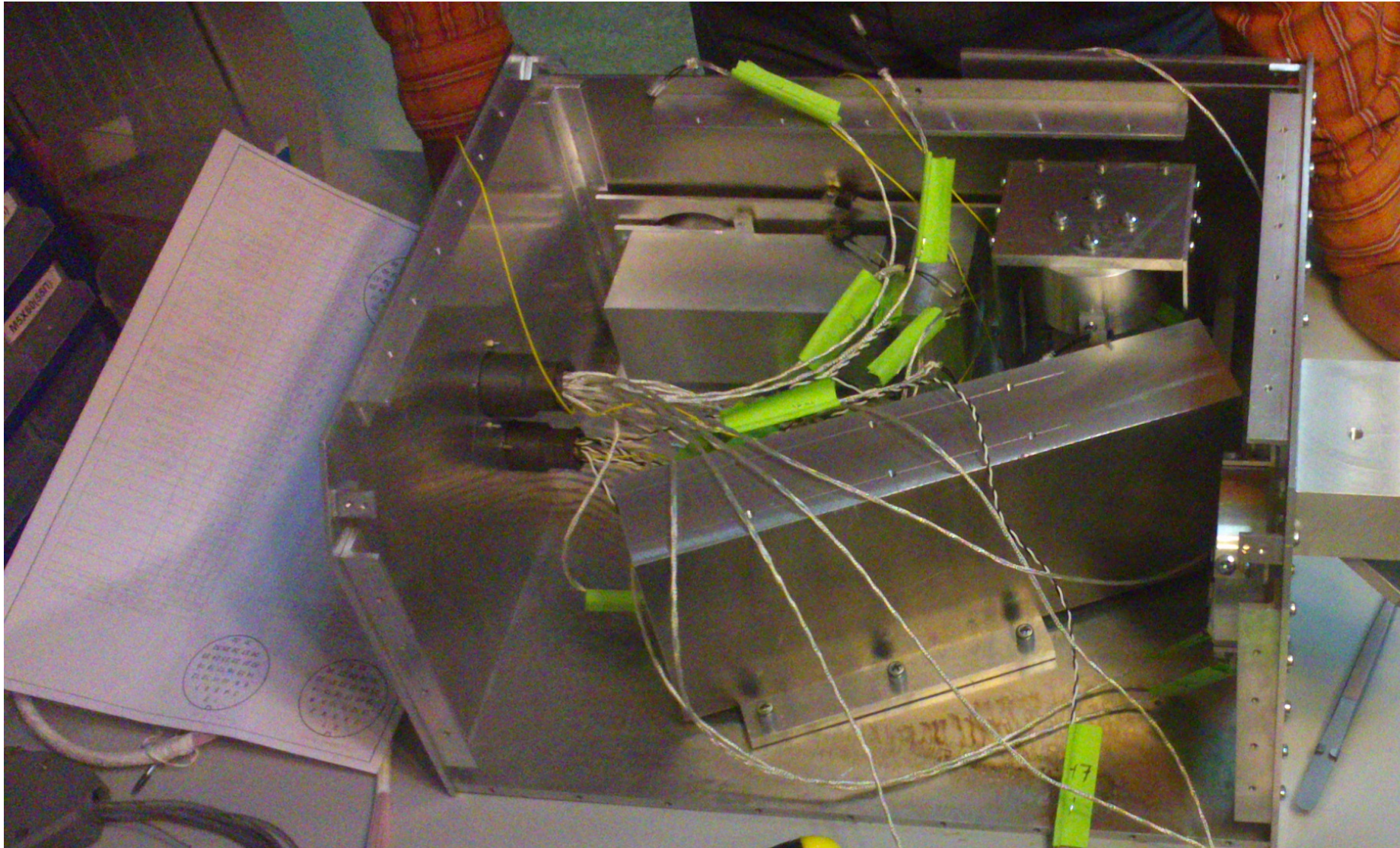
- 2 – Scanner module
- 4 – Blackbody unit
- 9 – Interferometer
- 10 – Stirling cooler
- 11 – ZnSe aspheric lens unit
- 12 – Dichroic & parabolic mirror unit



# HW: Thermal Model

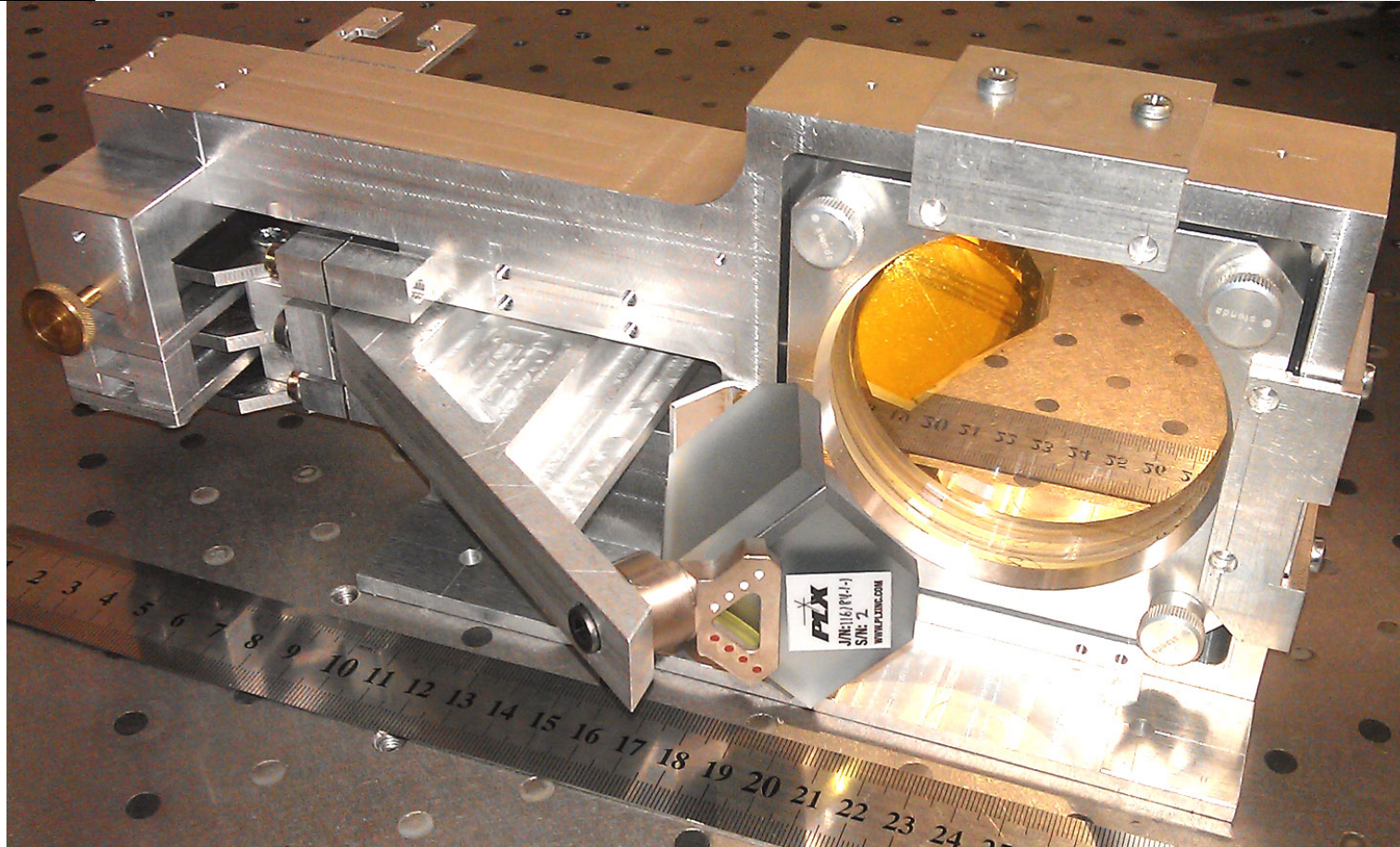


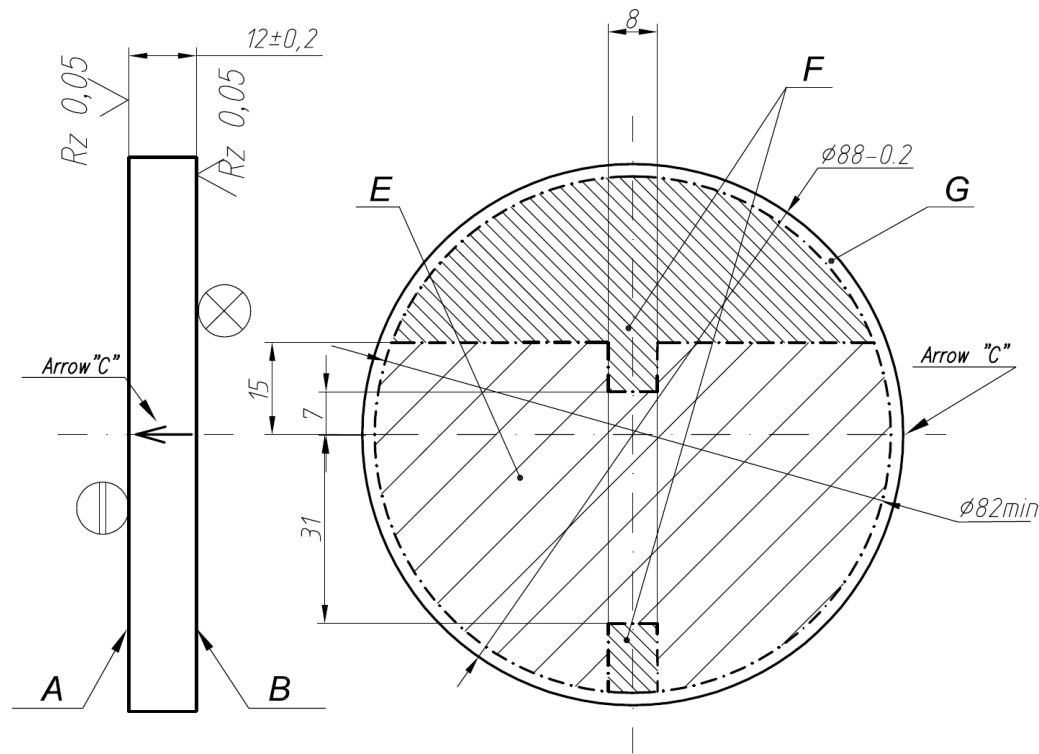
# HW: Thermal Model



TIRVIM/ACS (ExoMars-2016)

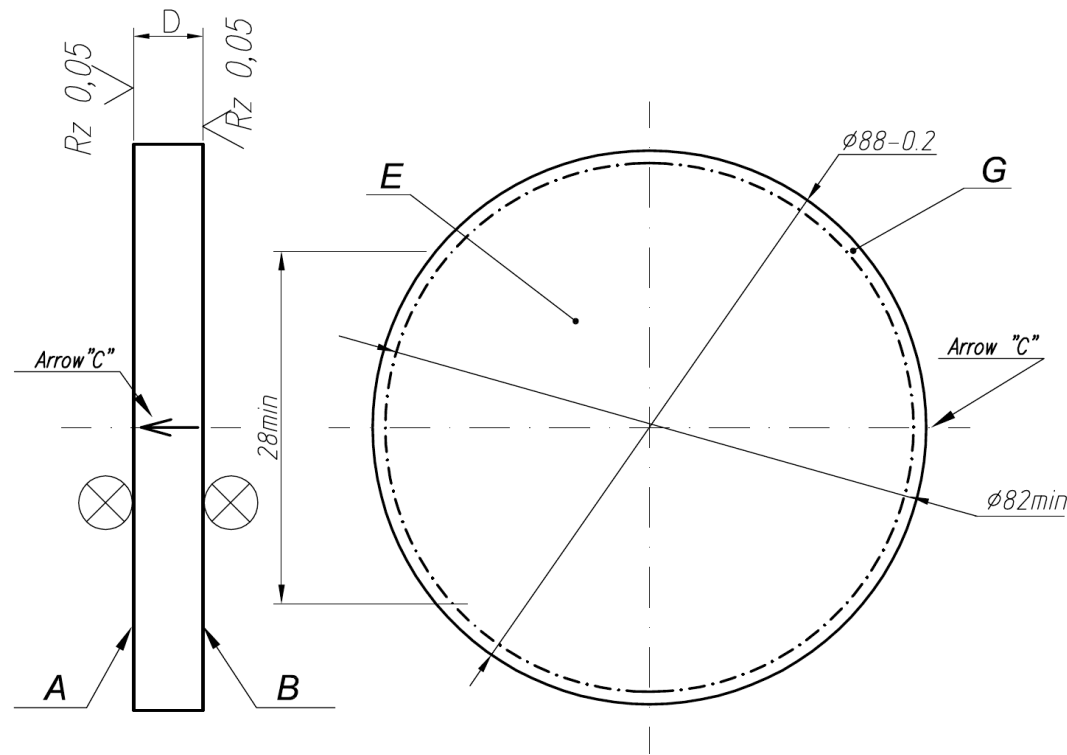
HW: Interferometer





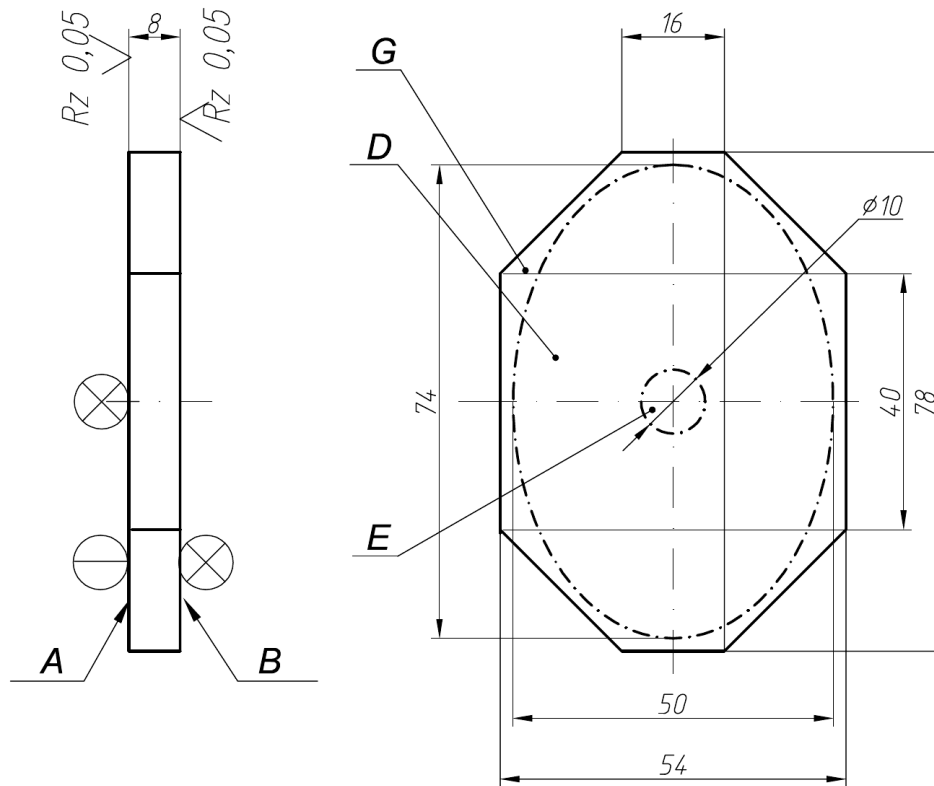
## BeamSplitter (preliminary drawing)

1. ALPHA (angle between planes A & B) =  $3 \pm 1$  arcminutes
2. Zones E & F may overlap by 0.5mm max
3. Planarity of planes A & B in zones E & F:  $N=0.7$ ,  $dN=0.3$
4. Painted arrow "C" indicates the most thick place with max error  $\pm 2$ mm
5.  $\textcircled{\ominus}$  - BeamSplitting coating:
  - in zone E:  $R=0.5 \pm 0.05$  for 3.3 & 15 micron;
  - $R=0.5 \pm 0.1$  for rest parts of ranges 3.1-3.5 & 14-16 micron;
  - $R=0.5 \pm 0.2$  in the rest parts of the region 2-17 micron.
  - in zone F:  $R=0.5 \pm 0.2$  for 0.76 & 0.63 micron.
6.  $\textcircled{\otimes}$  - antireflective coating:
  - $R < 0.02$  for 3.1-3.5 & 14-16 micron;
  - $R < 0.07$  in the rest parts of the region 2-17 micron;
  - $R < 0.1$  for 0.76 & 0.63 micron.
7. Reflection must be measured at Angle of Incidence  $35 \pm 5$  degrees.
8. Zone G is a non-working zone.
9. Bevels at sharp edges:  $0.5\text{mm} * 45$  degrees.



## Compensator (preliminary drawing)

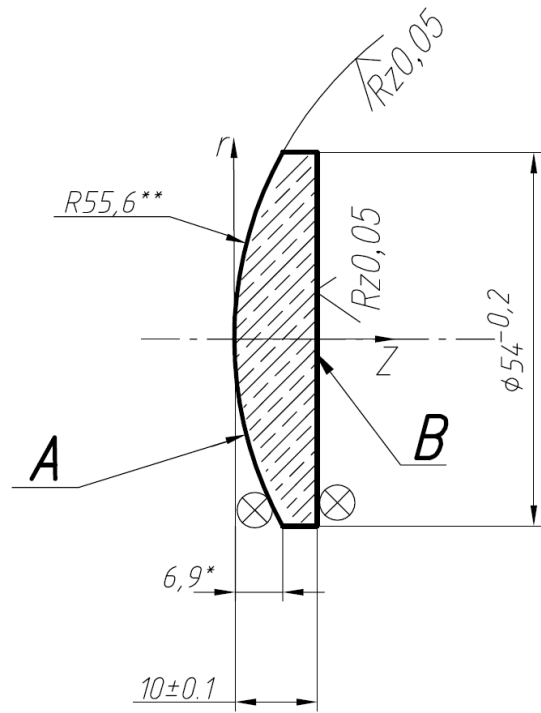
1. The angle between planes A & B must be equal to that of Beamsplitter ("ALPHA") with accuracy  $\pm 5arcseconds$
2. The Compensator thickness "D" must be equal to that of Beamsplitter with accuracy  $\pm 0.01mm$
3. Planarity of planes A & B in zone E:  $N=0.7$ ,  $dN=0.3$
4. Painted arrow "C" indicates the most thick place with max error  $\pm 2mm$
5.  $\otimes$  - antireflective coating:
  - $R < 0.02$  for 3.1-3.5 & 14-16 micron;
  - $R < 0.07$  in the rest parts of the region 2-17 micron
  - $R < 0.1$  at 0.76 & .63 micron.
6. Reflection must be measured at Angle of Incidence  $35 \pm 5degrees$ .
7. Zone G is a non-working zone.
8. Bevels at sharp edges:  $0.5mm * 45degrees$ .



## Dichroic

(preliminary drawing)

1. Angle between planes A & B:  $<5\text{arcminutes}$ .
2. Planarity of planes A & B in zones D & E:  $N=3$ ,  $dN=0.5$ .
3.  $\odot$  - dichroic coating in zone D:  
 at  $3.3\text{micron}$   $R>0.97$ ,  $T$  is not important;  
 in rest parts of range  $3.1\text{--}3.5\text{micron}$   $R>0.9$ ,  $T$  is not important;  
 in rest parts of range  $2\text{--}4\text{micron}$   $R>0.7$ ,  $T$  is not important;  
 at  $15\text{micron}$   $T>0.97$ ,  $R$  is not important;  
 in rest parts of range  $14\text{--}16\text{micron}$   $T>0.9$ ,  $R$  is not important;  
 in rest parts of range  $5\text{--}17\text{micron}$   $T>0.7$ ,  $R$  is not important.
4.  $\otimes$  - ARC in zone E of plane A and zones D & E of plane B:  
 $R<0.02$  in range  $14\text{--}16\text{micron}$ ;  
 $R<0.1$  in the rest parts of the region  $2\text{--}17\text{micron}$ .
5. Reflection must be measured at Angle of Incidence  $45\pm 5\text{degrees}$ .
6. Zone G is a non-working zone.
7. Bevels at sharp edges:  $0.5\text{mm} * 45\text{degrees}$ .



Однородность	1
Двулучепрел.	1
Бесшвильность	1A
Пузырность	1A
$N$	5
$\Delta N$	1
$P_A$	VI
$Z_F(\lambda=15\mu)$	42,8
$\phi_{св}$	50

## Aspheric ZnSe lens

(preliminary drawing)

1. Рабочая область 2...17 мкм.
2. \* – размер для справки.
3. Фаски на ребрах  $0,5\pm 0,2 \times 45^\circ$ .
4.  $\otimes$  – Просветл,  $R < 0,02$  при  $\lambda = (14...16)$  мкм.  $R < 0,07$  в остальных.

5. Уравнение поверхности A (гипербола):

$$Z = \frac{Gr^2}{1 + \sqrt{1 + 0,8G^2 r^2}} \quad G = 0,02$$

6. \*\* – радиус исходной описанной сферы.

(макс. разность  $Z_{сфер} - Z_{гипер}$  равна 0,24 мм при  $r = 18$  мм)

7.  $Z_F$  – координата фокуса по оси Z.



**FAST –  
Fourier for  
Atmospheric  
Species and  
Temperature**

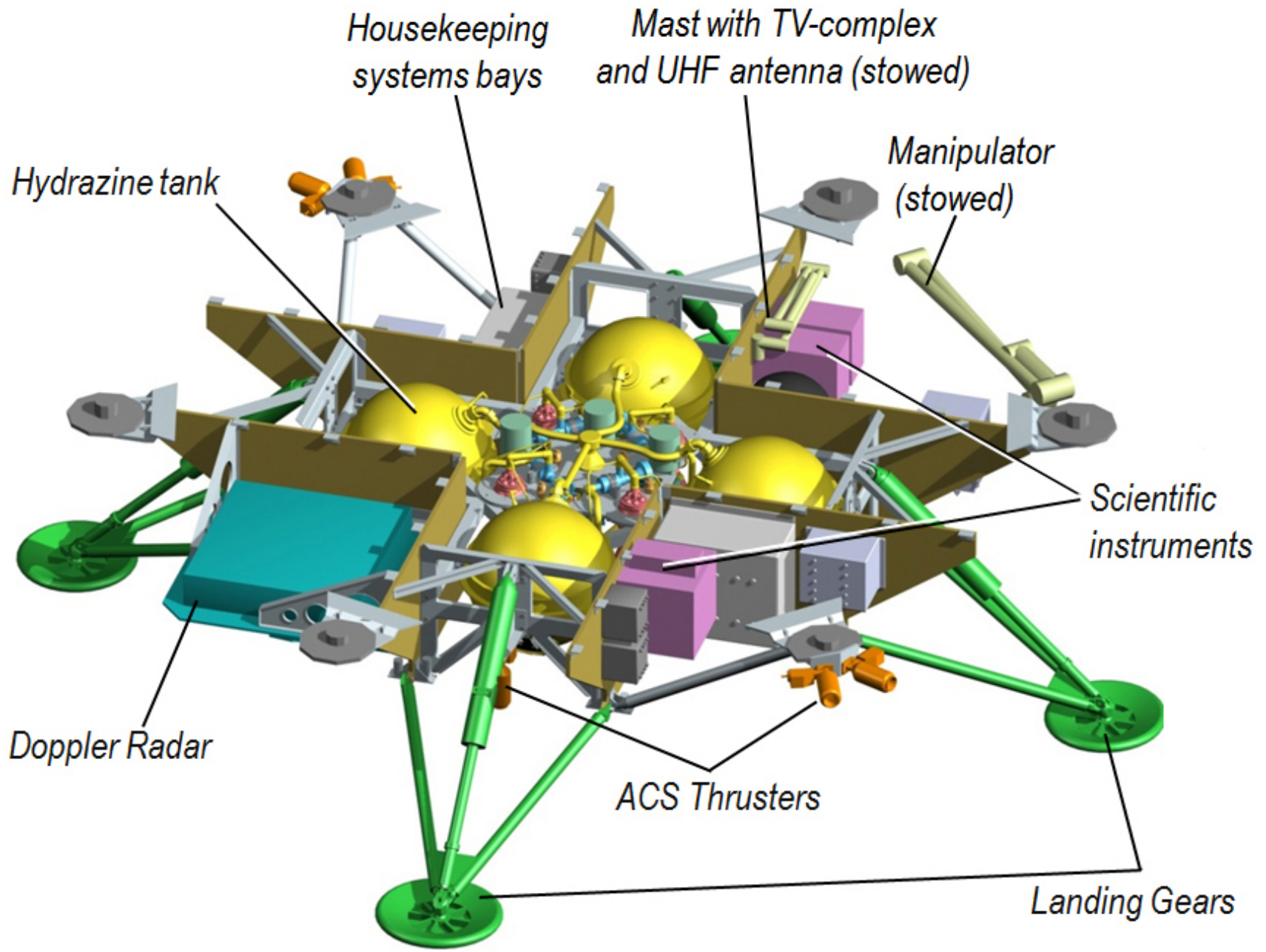
**(0.05cm<sup>-1</sup> !)**

FAST has won the contest:

FAST (first place in the list):	76 points
Next after FAST in the list:	68 points
The last in the list :	30 points

# Landing platform

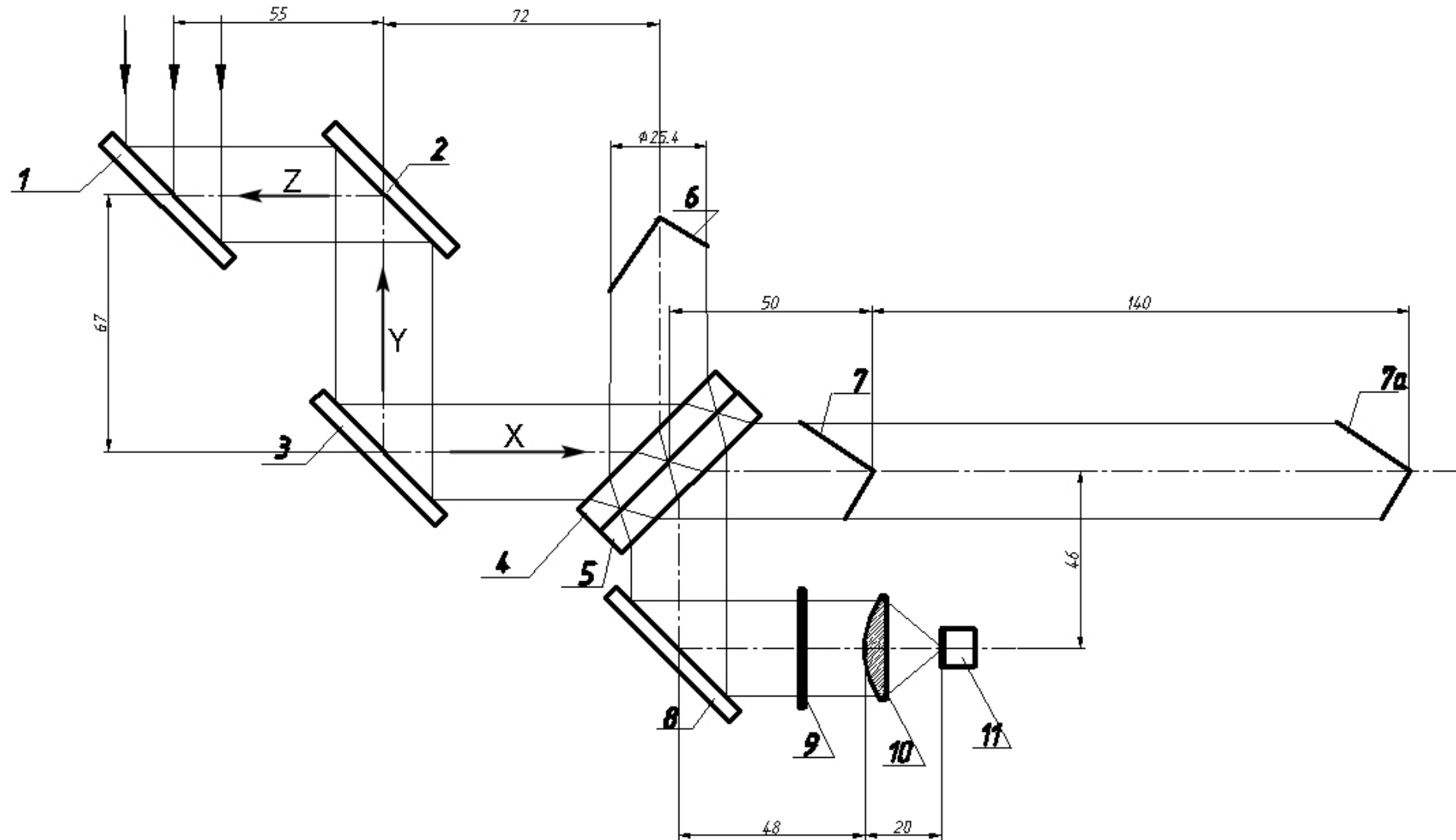
**FAST (ExoMars-2018)**



# Main scientific objectives

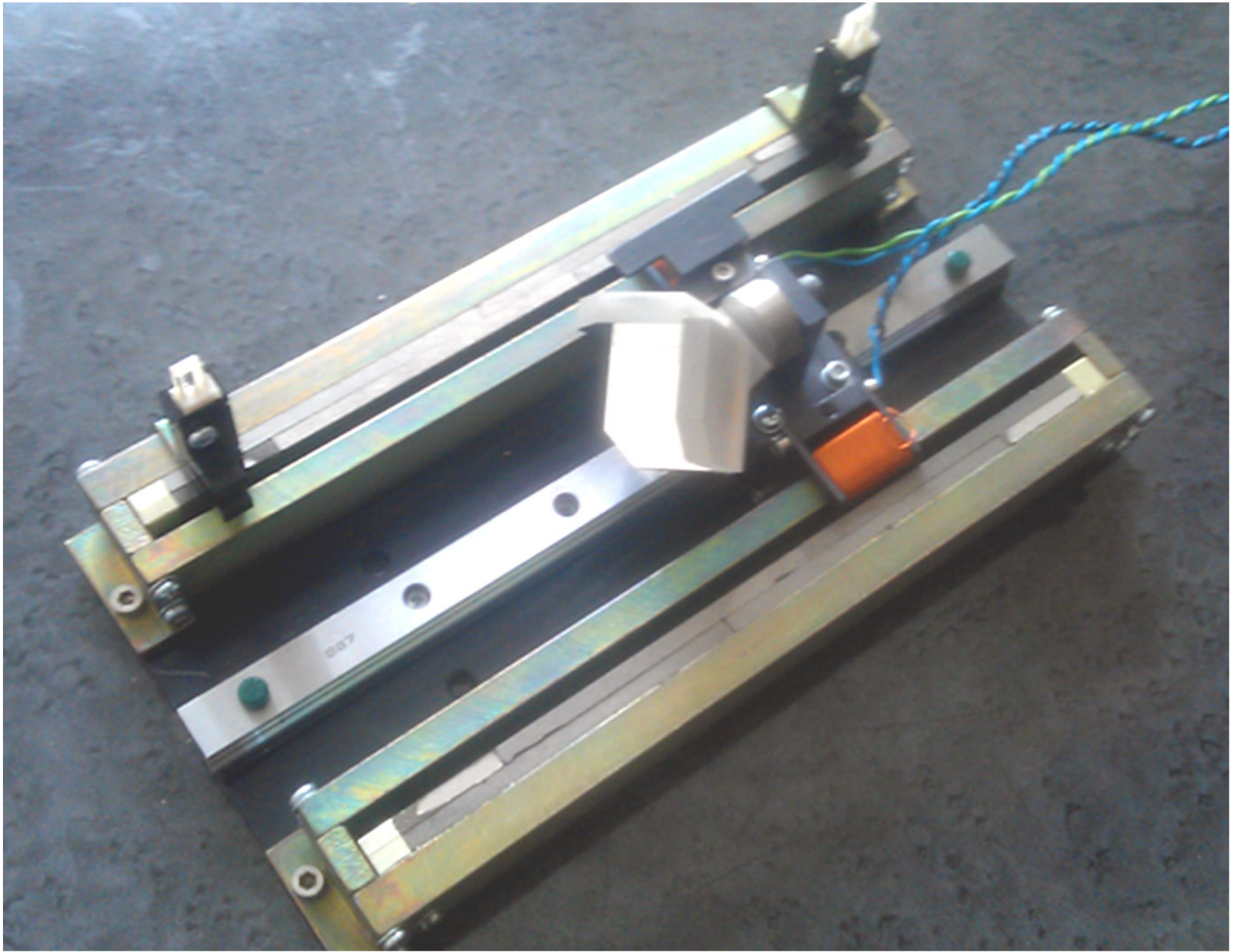
OBJECTIVE	MESUREMENT MODES & PARAMETERS
Methane & other minor atmosphere constituents	Sun tracking, $0.05\text{cm}^{-1}$ 1) 2-4 $\mu\text{m}$ , PV-PbCdSe detector @200K, 1 IFG: 10sec or: 2) Pyro-detector @RT, 1 IFG: 30sec
Vertical thermal profile of the atmosphere, both day & night	Different air masses, $\text{CO}_2$ band at 15 $\mu\text{m}$ , $1.6\text{cm}^{-1}$ Pyro-detector @RT, 1 IFG: 30sec, averaging
Mineral composition of the surroundings	1) 2-4 $\mu\text{m}$ , PV-PbCdSe detector @200K, 1 IFG: 10sec or: 2) Pyro-detector @RT, 1 IFG: 30sec, averaging

## Concept



# Carriage unit prototype

FAST (ExoMars-2018)

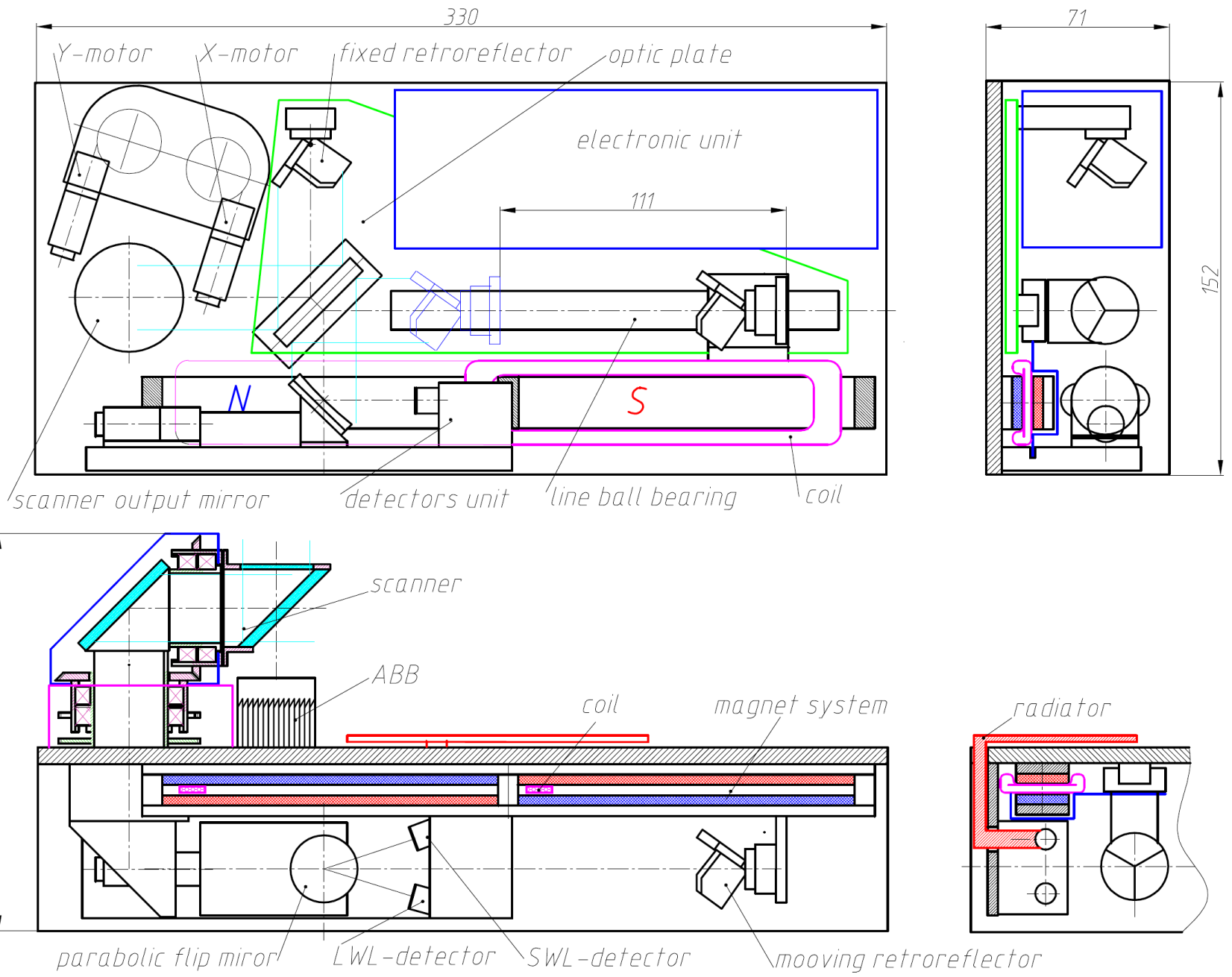


**Size mockup**



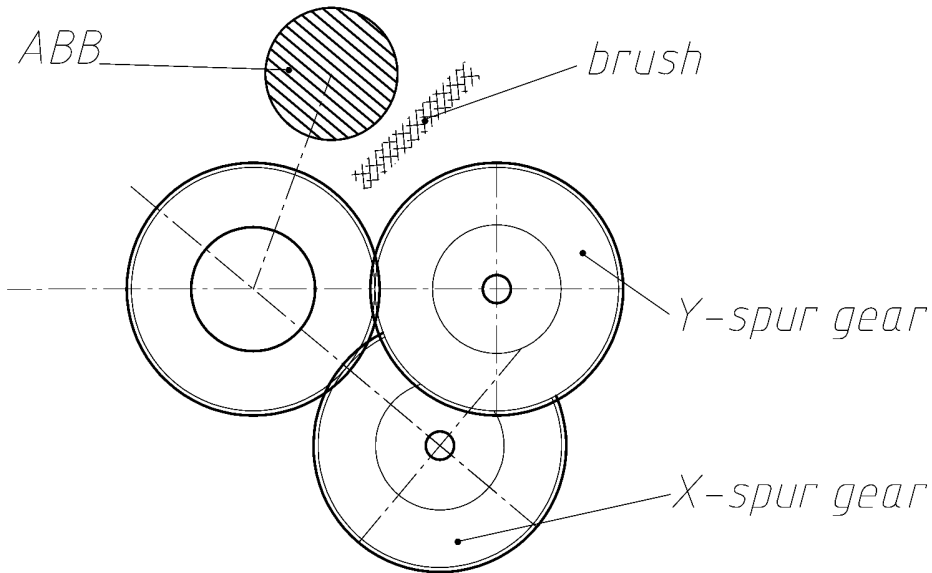
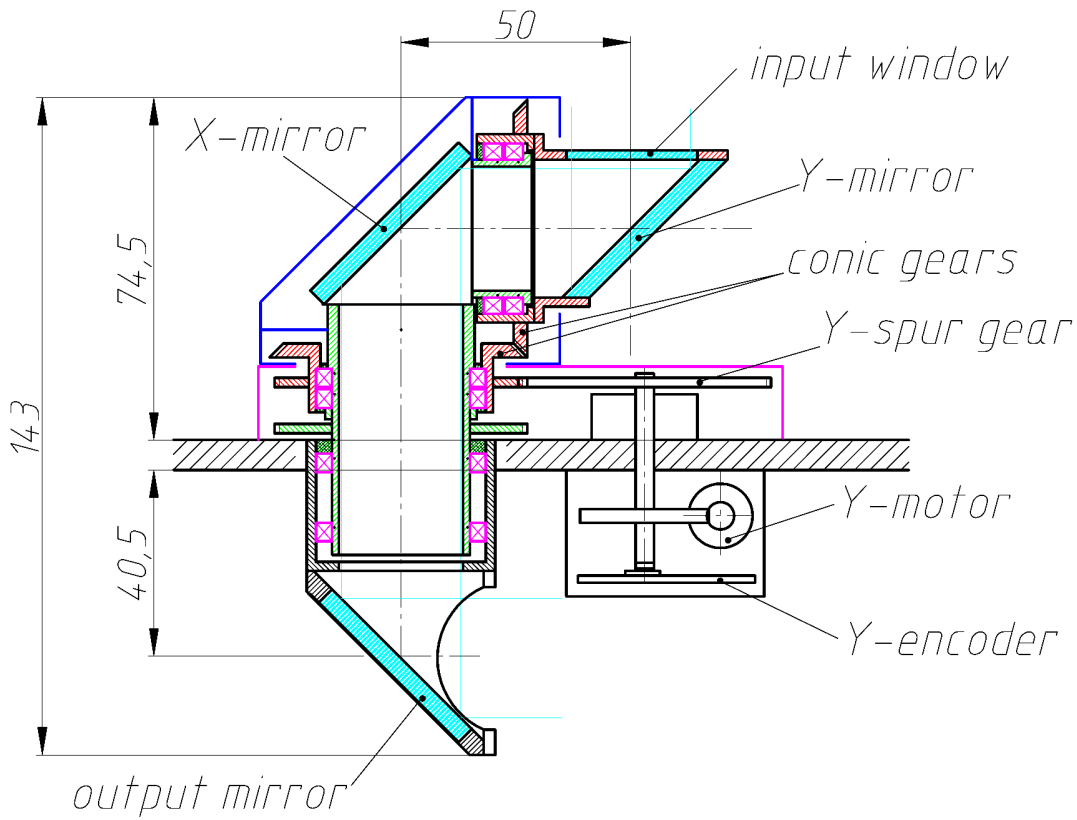
# Layout

# FAST (ExoMars-2018)





# Scanner layout



# Possible German inputs for FAST

<p><b>Moving mirror system</b></p>	<p>Rail with carriage, motor and electronics,incl. PID system, providing the stable 11-cm movement</p>
<p><b>Key optical subsystems:</b> design (together with IKI), manufacturing (procurement), tests, measurements, certification,...</p>	<ul style="list-style-type: none"> <li>• <b>Beamsplitter-Compensator unit (ZnSe)</b></li> <li>• <b>Laser for reference channel</b></li> <li>• <b>Au-coated metallic mirrors for Scanner</b></li> <li>• ...</li> <li>• <b>Retro-reflectors (1" cube corners)</b></li> <li>• ...</li> </ul>
<p><b>Manufacturing of mechanical components</b></p>	<p>According to drawings issued by IKI and ASTROFEIN,...</p>
<p>On-board BlackBody</p>	<p>Design, manufacturing, tests, certification,...</p>
<p>...?</p>	